

In re Patent Application of:
WRIGHT ET AL.
Serial No. **09/976,647**
Filed: **October 11, 2001**

REMARKS

Claims 59-75 remain in this application. Claims 59, 65, 68, 69 and 75 have been amended.

Applicants file this Amendment Under 37 CFR §1.607 with existing claims 59-75 that had been added and copied verbatim from U.S. Patent No. 6,181,990, granted January 30, 2001, to John Francis Grabowsky and David Ray Stevens (hereinafter "Grabowsky") for purposes of provoking an interference with that patent. The U.S. Patent and Trademark Office considers these amended claims patentable based on a reexamination of these amended claims in a reexamination of the Grabowsky patent, filed on August 12, 2003 as application control no. 90/006,742.

In an Office Action dated August 9, 2002, the Examiner rejected the originally copied claims from Grabowsky as being unpatentable, and thus, an interference could not be initiated since a prerequisite for interference under 37 CFR §1.606 is that the claim be patentable to the applicant subject to a judgment in the interference. Original claims 59, 62-70 and 75 were rejected as anticipated by U.S. Patent No. 5,351,194 to Ross et al. (hereinafter "Ross") and other claims as obvious over Ross in view of U.S. Patent No. 5,652,717 to Miller et al. (hereinafter "Miller"), U.S. Patent No. 5,943,399 to Bannister et al. (hereinafter "Bannister"), or Ross in view of U.S. Patent No. 5,463,656 to Polivka et al. (hereinafter "Polivka").

A request for reexamination of Grabowsky was filed on August 12, 2003 as application control no. 90/006,742. During the reexamination, the patent owner amended independent claims to overcome the rejections over the cited prior art.

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In the reasons for patentability/confirmation mailed October 3, 2005, the Examiner stated that claims 1-51 of Grabowsky are patentable over the prior art of record. According to the Examiner, as argued by the patent owner, the art of record failed to teach an aircraft data transmission system and method comprising, among other limitations, at least one first sensor on the aircraft which gathers in-flight data and at least one second sensor configured to sensing a landing of the aircraft, wherein communication is initiated via a cellular infrastructure in response to the second sensor sensing the landing of the aircraft.

Applicants have amended claims 59, 65, 68, 69 and 75 in the manner as allowed in the reexamination of Grabowsky. Applicants' disclosure specifically recites a plurality of transducers as set forth in the claim chart, corresponding to at least first and second sensors.

Grabowsky also amended the independent claims to include the recitation that the flight data includes time, airspeed, altitude, vertical acceleration, and heading data relating to a flight of the aircraft. Applicants note that Federal Aviation Administration Section 121-343 (1994) mandates that large airplanes certified for operation above 25,000 feet or turbine-engine powered must be equipped with one or more approved flight recorders that record data relating to time, altitude, airspeed, vertical acceleration and heading. Other data are also recorded as set forth in the regulation. Thus, any flight data acquired by the DFDAU and DFDR inherently includes this data. These parameters are directly from the FAA requirements for "black boxes," i.e., the flight data recorders, and a copy of section 121.343 is

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enclosed with this Amendment as Exhibit 1. This FAA requirement is dated May 24, 1994, which predates the effective filing date of November 14, 1995 for the instant application and the July 30, 1998 filing date of Grabowsky.

Claims 59-62 correspond to Claims 1-4 of Grabowsky. Claims 63 and 64 correspond to Claims 6 and 7 of Grabowsky. Claims 65-74 correspond to Claims 15-24 of Grabowsky. Claim 75 corresponds to Claim 33 of Grabowsky.

In accordance with 37 CFR §1.607(a), the copied claims may be specifically applied to Applicants' disclosure as follows:

Copied Claim	Applicants' Disclosure
59. An aircraft data transmission system,	Title: page 1, lines 1-2
the aircraft having a data acquisition unit,	DFDAU 16, page 20, lines 16-26; DFDR 18 operative with GDL 101, page 21, line 11-17.
and the aircraft including a data storage medium having stored thereon flight data gathered in-flight by at least a first sensor on the aircraft, comprising:	GDL data storage and communications unit 111 (FIG. 3) stores flight data. GDL unit synchronizes with the flight parameter data stream from the DFDAU 16, and stores the collected data in memory. Page 20, lines 1-26. There are a plurality of sensors, described as aircraft flight parameter transducers, Page 20, lines 11-15. FIG. 2.
a communications unit located in the aircraft and in communication with the data acquisition unit;	GDL airborne segment 101, GDL unit 111, GDL antenna 113, page 16, lines 19-22; page 20, lines 16-22.
at least a second sensor configured to sense landing of the aircraft	The aircraft data is provided by the airborne data acquisition unit in a compressed and

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a cellular infrastructure in communication with said communications unit after the aircraft has landed, wherein the cellular infrastructure communicates said flight data, and

wherein the communication is initiated when at least the second sensor senses the landing of the aircraft;

a data reception unit in communication with said cellular infrastructure; and

wherein said flight data includes time, airspeed, altitude, vertical acceleration, and heading data relating to a flight of the aircraft.

60. The system of claim 59

Applicants' Disclosure

encrypted format that is automatically downloaded to an airport-resident base station segment when the aircraft lands. Page 41, lines 5-10. A second transducer, i.e., second sensor, must sense the landing of the aircraft in order to download automatically any data.

FIG. 1A, circular cells defined by wireless routers 201 and base stations 202; FIG. 4, circular cells 214, 215; page 15, lines 23-24; page 23, lines 21-23. Page 37, line 20-25 defines the system as cellular infrastructure typical of cellular telephone network. Data is communicated through this network to the server/archive 204, 304. Page 17, lines 18-23.

Page 41, lines 7-9; "that is automatically downloaded ... when aircraft lands." Second sensor, i.e., transducer, must be used.

Server/archive 204 in association with server/archive 304; page 17, lines 18-23.

Federal Aviation Administration Section 121-343 (1994) mandates that large airplanes that fly above 25,000 feet and turbo-engine powered airplanes must be equipped with one or more approved flight recorders, and must record data relating time, altitude, airspeed, vertical acceleration and heading. Thus, the flight data acquired by the DFDAU and DFDR inherently includes this listed data.

Transmission Control Protocol/

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wherein said data reception unit is in communication with said cellular infrastructure via the Internet.

61. The system of claim 59 wherein said data reception unit is in communication with said cellular infrastructure via the public switch telephone network.

62. The system of claim 59 wherein said communications unit has at least one modem in communication with said cellular infrastructure and

said data reception unit has at least one modem in communication with said cellular infrastructure.

63. The system of claim 59 wherein said cellular infrastructure includes:

an antenna;

a transceiver subsystem in communication with said antenna; and

a controller in communication with said transceiver subsystem.

64. The system of claim 59 wherein said data reception unit includes:

a router; and

Applicants' Disclosure

Internet Protocol (TCP/IP) operative in Ethernet LAN 207 with TELCO connection (FIG. 1). Clearly defined use with Internet.

Server/archive 304, gateway segment 306 in communication with ground subsystem 200 via ISDN TELCO (FIG. 1); page 18, lines 6-9. TELCO is public switch telephone network.

Network transceiver 26 naturally includes modem to modulate/demodulate signals.

Base station 202 naturally includes modem with server 204 to demodulate/modulate signals and operative with Ethernet LAN 207.

Antenna 222, 223, FIG. 5, page 25, lines 18-23.

Transceiver 221, FIG. 5, page 25, lines 18-23.

Controller/processor 225, FIG. 5, page 26, lines 3-6.

Router 201

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a processor in communication with
said router,

said processor having a storage
unit.

65. An aircraft data transmission
system, the aircraft having a
data acquisition unit,

the aircraft including a data
storage medium having stored
thereon flight data gathered in-
flight by at least one sensor on
the aircraft, comprising:

sensing means for sensing a
landing of the aircraft;

means for transmitting said
flight data from the data
acquisition unit, via a cellular
infrastructure after the aircraft
has landed,

Applicants' Disclosure

Server 304 in communication with
router 201, FIG. 1; page 19,
lines 5-12.

Archive includes memory, database
management software; page 19,
lines 5-12.

Title: page 1, lines 1-2

DFDAU 16, page 20, lines 16-26;
DFDR 18 operative with GDL 101,
page 21, line 11-17.

GDL data storage and
communications unit 111 (FIG. 3)
stores flight data. GDL unit
synchronizes with the flight
parameter data stream from the
DFDAU 16, and stores the
collected data in memory. Page
20, lines 1-26. There are a
plurality of sensors, described
as aircraft flight parameter
transducers, Page 20, lines
11-15. FIG. 2.

The aircraft data is provided by
the airborne data acquisition
unit in a compressed and
encrypted format that is
automatically downloaded to an
airport-resident base station
segment when the aircraft lands.
Page 41, lines 5-10. A second
transducer, i.e., second sensor,
must sense the landing of the
aircraft in order to download
automatically any data.

GDL airborne segment 101, GDL
unit 111, GDL antenna 113, page
16, lines 19-22; page 20, lines
16-22. FIG. 1A, circular cells
defined by wireless routers 201
and base stations 202; FIG. 4,
circular cells 214, 215; page 15,

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Applicants' Disclosure

wherein transmission of the data is initiated when the sensing means sense the landing of the aircraft;

means for receiving said flight data from said cellular infrastructure; and

wherein said flight data includes time, airspeed, altitude, vertical acceleration, and heading data relating to a flight of the aircraft.

66. The system of claim 65 wherein said means for transmitting data includes a processor.

67. The system of claim 65 wherein said means for receiving data includes a processor.

68. A method of transmitting aircraft flight data from an aircraft, comprising:

receiving flight data from a data acquisition unit;

lines 23-24; page 23, lines 21-23. Page 37, line 20-25 defines the system as cellular infrastructure typical of cellular telephone network.

Page 41, lines 7-9; "that is automatically downloaded . . . when aircraft lands." Sensor must be used.

Server/archive 204 in association with server/archive 304; page 17, lines 18-23.

Federal Aviation Administration Section 121-343 (1994) mandates that large airplanes that fly above 25,000 feet and turbo-engine powered airplanes must be equipped with one or more approved flight recorders, and must record data relating time, altitude, airspeed, vertical acceleration and heading. Thus, the flight data acquired by the DFDAU and DFDR inherently includes this listed data.

Processor 22, FIG. 3.

Server 304 in communication with router 201, FIG. 1; page 19, lines 5-12.

Title: page 1, lines 1-2

DFDAU 16, page 20, lines 16-26; DFDR 18 operative with GDL 101, page 21, line 11-17.

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receiving a signal indicating a landing of the aircraft from at least a first sensor;

transmitting said flight data via a cellular communications infrastructure after the aircraft has landed,

wherein the cellular communications infrastructure is accessed in response to the signal; and

receiving said transmitted flight data, and

wherein said flight data is gathered in-flight by at least a second sensor on the aircraft, and includes time, airspeed, altitude, vertical acceleration, and heading data relating to a flight of the aircraft.

Applicants' Disclosure

The aircraft data is provided by the airborne data acquisition unit in a compressed and encrypted format that is automatically downloaded to an airport-resident base station segment when the aircraft lands. Page 41, lines 5-10. A transducer, i.e., first sensor, must sense the landing of the aircraft in order to download automatically any data.

FIG. 1A, circular cells defined by wireless routers 201 and base stations 202; FIG. 4, circular cells 214, 215; page 15, lines 23-24; page 23, lines 21-23. Page 37, line 20-25 defines the system as cellular infrastructure typical of cellular telephone network.

Page 41, lines 7-9; "that is automatically downloaded . . . when aircraft lands." Thus, data is transmitted through the cellular communication infrastructure.

Server/archive 204 in association with server/archive 304; page 17, lines 18-23.

GDL data storage and communications unit 111 (FIG. 3) stores flight data. GDL unit synchronizes with the flight parameter data stream from the DFDAU 16, and stores the collected data in memory. Page 20, lines 1-26. There are a plurality of sensors, described as aircraft flight parameter transducers, Page 20, lines 11-15. FIG. 2. Federal Aviation Administration Section 121-343

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69. A computer-implemented method of transmitting aircraft flight data from an aircraft, comprising:

receiving flight data from a digital flight data acquisition unit;

wherein said flight data is gathered in-flight by at least a first sensor on the aircraft, and includes time, airspeed, altitude, vertical acceleration, and heading data relating to a flight of the aircraft;

Applicants' Disclosure

(1994) mandates that large airplanes that fly above 25,000 feet and turbo-engine powered airplanes must be equipped with one or more approved flight recorders, and must record data relating time, altitude, airspeed, vertical acceleration and heading. Thus, the flight data acquired by the DFDAU and DFDR inherently includes this listed data.

Title: page 1, lines 1-2.

DFDAU 16, page 20, lines 16-26;
DFDR 18 operative with GDL 101, page 21, line 11-17.

GDL data storage and communications unit 111 (FIG. 3) stores flight data. GDL unit synchronizes with the flight parameter data stream from the DFDAU 16, and stores the collected data in memory. Page 20, lines 1-26. There are a plurality of sensors, described as aircraft flight parameter transducers, Page 20, lines 11-15. FIG. 2. Federal Aviation Administration Section 121-343 (1994) mandates that large airplanes that fly above 25,000 feet and turbo-engine powered airplanes must be equipped with one or more approved flight recorders, and must record data relating time, altitude, airspeed, vertical acceleration and heading. Thus, the flight data acquired by the DFDAU and DFDR inherently includes this listed data.

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receiving a signal indicating a landing of the aircraft from at least a second sensor.

processing said flight data to prepare said data for transmission; and

transmitting said processed data via a cellular infrastructure after the aircraft has landed,

wherein the cellular infrastructure is accessed in response to the signal.

70. The method of claim 69 further comprising receiving said transmitted data at a flight operations center.

Applicants' Disclosure

The aircraft data is provided by the airborne data acquisition unit in a compressed and encrypted format that is automatically downloaded to an airport-resident base station segment when the aircraft lands. Page 41, lines 5-10. A second transducer, i.e., second sensor, must sense the landing of the aircraft in order to download automatically any data.

GDL airborne segment 101, GDL unit 111, GDL antenna 113, page 16, lines 19-22; page 20, lines 16-22.

FIG. 1A, circular cells defined by wireless routers 201 and base stations 202; FIG. 4, circular cells 214, 215; page 15, lines 23-24; page 23, lines 21-23. Page 37, line 20-25 defines the system as cellular infrastructure typical of cellular telephone network.

Page 41, lines 7-9; "that is automatically downloaded . . . when aircraft lands." Thus, data is transmitted through the cellular communications infrastructure.

FIG. 1 Remote Flight Operations Control Center 300.

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71. The method of claim 70 further comprising receiving said transmitted data and transmitting said received data via the Internet before receiving said transmitted data at a flight operations center.

72. The method of claim 70 further comprising receiving said transmitted data and

transmitting said received data via the public-switched telephone network before receiving said transmitted data at a flight operations center.

73. The method of claim 69 wherein processing said flight data includes:

compressing said flight data;

encrypting said flight data;

segmenting said flight data; and

constructing packets of data from said segmented flight data.

74. The method of claim 69 wherein receiving said transmitted data includes:

acknowledging receipt of said transmitted data;

Applicants' Disclosure

Transmission Control Protocol/Internet Protocol (TCP/IP) operative in Ethernet LAN 207 with TELCO connection (FIG. 1). Clearly defined use with Internet.

Server/archive 304, gateway segment 306 in communication with ground subsystem 200 via ISDN TELCO (FIG. 1); page 18, lines 6-9.

Source coding can be used for data compression. Aircraft data downloaded as compressed data. Page 27, lines 9-12 and line 25.

Aircraft flight data is encrypted. Page 27, line 10.

Flight data is segmented into channels. Flight data is multiplexed. Page 27, lines 9 and 19-20.

TCP/IP is packet protocol. FIG 1. System produces "flight performance data packet." Page 31, line 10, page 32, line 2.

Polling occurs and receipts of packets acknowledged and retransmissions requested when

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reassembling said received data;
decrypting said reassembled data;

uncompressing said decrypted
data; and

storing said uncompressed data.

75. A computer readable medium
having stored thereon
instructions which when executed
by a processor, cause the
processor to perform the steps
of:

receiving flight data from a
digital flight data acquisition
unit in an aircraft;

wherein said flight data is
gathered in-flight by at least a
first sensor on the aircraft, and
includes time, airspeed,
altitude, vertical acceleration,
and heading data relating to a
flight of the aircraft;

Applicants' Disclosure

errors occur. Standard use of
TCP/IP. FIG. 1. Page 9, lines
1-24. Page 41, lines 9-13.

FIG. 1. Base station segment
operative with wireless bridge
segment and receives packets
based on TCP/IP and operative
with remote flight operations
control center 300; Also
operative with GDL work station
segment 303 and controller 301 to
acknowledge receipt, reassemble
data, decrypt, uncompress and
store for further use in
server/archive 304.

GDL unit includes processor 22
(FIG. 3) associated with memory
24 as stored instructions.

Title: page 1, lines 1-2

DFDAU 16, page 20, lines 16-26;
DFDR 18 operative with GDL 101,
page 21, line 11-17.

GDL data storage and
communications unit 111 (FIG. 3)
stores flight data. GDL unit
synchronizes with the flight
parameter data stream from the
DFDAU 16, and stores the
collected data in memory. Page
20, lines 1-26. There are a
plurality of sensors, described
as aircraft flight parameter
transducers, Page 20, lines 11-
15. FIG. 2. Federal Aviation
Administration Section 121-343

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receiving a signal indicating a landing of the aircraft from at least a second sensor.

processing said flight data to prepare said data for transmission; and

transmitting said processed data via a cellular infrastructure when said aircraft has landed,

wherein the cellular infrastructure is accessed in response to the signal.

Applicants' Disclosure

(1994) mandates that large airplanes that fly above 25,000 feet and turbo-engine powered airplanes must be equipped with one or more approved flight recorders, and must record data relating time, altitude, airspeed, vertical acceleration and heading. Thus, the flight data acquired by the DFDAU and DFDR inherently includes this listed data.

The aircraft data is provided by the airborne data acquisition unit in a compressed and encrypted format that is automatically downloaded to an airport-resident base station segment when the aircraft lands. Page 41, lines 5-10. A second transducer, i.e., second sensor, must sense the landing of the aircraft in order to download automatically any data.

GDL airborne segment 101, GDL unit 111, GDL antenna 113, page 16, lines 19-22; page 20, lines 16-22.

FIG. 1A, circular cells defined by wireless routers 201 and base stations 202; FIG. 4, circular cells 214, 215; page 15, lines 23-24; page 23, lines 21-23. Page 37, line 20-25 defines the system as cellular infrastructure typical of cellular telephone network.

Page 41, lines 7-9; "that is automatically downloaded . . . when aircraft lands." Thus, data is transmitted through the cellular communications infrastructure.

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Applicants have amended verbatim from reexamined U.S. Patent No. 6,181,990 the allowed independent claims 1, 15, 18, 19, 20 and 33. Applicants note that these claim recitations are specifically disclosed in Applicants' original disclosure filed November 14, 1995. No affidavits or declarations under 37 CFR §1.608 are submitted with this Amendment because Applicants are the senior party.

Clearly, Applicants' Amendment shows that Applicants were in possession of the invention as now claimed in this Amendment. The chart applying the copied claims to Applicants' disclosure shows the use of an aircraft data acquisition unit that acquires data from throughout the aircraft. This data is downloaded after the plane lands at the airport. This can occur automatically after landing. The communication system is structured as a cellular infrastructure, as clearly seen in the cellular cells defined by wireless routers and operative with the TCP/IP protocol and ISDN TELCO as a public switch telephone network. The channel sharing and other communication system is akin to cellular telephone networks as clearly described in the specification. The data is received and processed at a remote flight operations control center. At least first and second sensors, i.e., "aircraft flight parameter transducers" are included. The listing of flight data is inherent to the DFDAU and DFDR used on airplanes according to FAA regulations.

Pursuant to 37 CFR §1.607, Applicants "present" the following proposed Count I based on amended claim 59:

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59. An aircraft data transmission system, the aircraft having a data acquisition unit and the aircraft including a data storage medium having stored thereon flight data gathered in-flight by at least a first sensor on the aircraft, comprising:

a communications unit located in the aircraft and in communication with the data acquisition unit;

at least a second sensor configured to sense a landing of the aircraft;

a cellular infrastructure in communication with said communications unit after the aircraft has landed, wherein the cellular infrastructure communicates said flight data, and wherein the communication is initiated when at least the second sensor senses the landing of the aircraft;

a data reception unit in communication with said cellular infrastructure; and

wherein said flight data includes time, airspeed, altitude, vertical acceleration, and heading data relating to a flight of the aircraft.

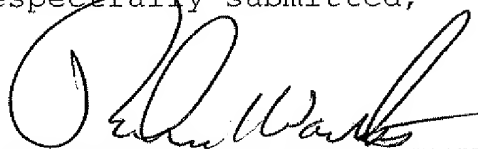
Applicants submit that the proposed Count I corresponds to patentees' Claims 1-4, 6, 7, 15-24 and 33 and Applicants' Claims 59-75 as amended.

Because the subject application claims original priority as a series of continuation applications to the parent application filed on November 14, 1995, which is about two years and eight months before the filing date of July 30, 1998 for the application that matured into U.S. Patent No. 6,181,990 to Grabowsky et al., Applicants are the senior party and no declarations under 37 CFR \$1.608 are required.

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Applicants respectfully request that an interference
be declared with Applicants as senior party.

Respectfully submitted,



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Sec. 121.343 - Flight recorders.

(a) Except as provided in paragraphs (b), (c), (d), (e), and (f) of this section, no person may operate a large airplane that is certificated for operations above 25,000 feet altitude or is turbine-engine powered unless it is equipped with one or more approved flight recorders that record data from which the following may be determined within the ranges, accuracies, and recording intervals specified in appendix B of this part:

- (1) Time;
- (2) Altitude;
- (3) Airspeed;
- (4) Vertical acceleration;
- (5) Heading; and
- (6) Time of each radio transmission either to or from air traffic control.

(b) No person may operate a large airplane type certificated up to and including September 30, 1969, for operations above 25,000 feet altitude, or a turbine-engine powered airplane certificated before the same date, unless it is equipped before May 26, 1989 with one or more approved flight recorders that utilize a digital method of recording and storing data and a method of readily retrieving that data from the storage medium. The following information must be able to be determined within the ranges, accuracies, and recording intervals specified in appendix B of this part:

- (1) Time;
- (2) Altitude;
- (3) Airspeed;
- (4) Vertical acceleration;
- (5) Heading; and

(6) Time of each radio transmission either to or from air traffic control.

(c) Except as provided in paragraph (l) of this section, no person may operate an airplane specified in paragraph (b) of this section unless it is equipped, before May 26, 1994, with one or more approved flight recorders that utilize a digital method of recording and storing data and a method of readily retrieving that data from the storage medium. The following information must be able to be determined within the ranges, accuracies and recording intervals specified in appendix B of this part:

(1) Time;

(2) Altitude;

(3) Airspeed;

(4) Vertical acceleration;

(5) Heading;

(6) Time of each radio transmission either to or from air traffic control;

(7) Pitch attitude;

(8) Roll attitude;

(9) Longitudinal acceleration;

(10) Control column or pitch control surface position; and

(11) Thrust of each engine.

(d) No person may operate an airplane specified in paragraph (b) of this section that is manufactured after May 26, 1989, as well as airplanes specified in paragraph (a) of this section that have been type certificated after September 30, 1969, unless it is equipped with one or more approved flight recorders that utilize a digital method of recording and storing data and a method of readily retrieving that data from the storage medium. The following information must be able to be determined within the ranges, accuracies, and recording intervals specified in appendix B of this part:

(1) Time;

(2) Altitude;

- (3) Airspeed;
- (4) Vertical acceleration;
- (5) Heading;
- (6) Time of each radio transmission either to or from air traffic control;
- (7) Pitch attitude;
- (8) Roll attitude;
- (9) Longitudinal acceleration;
- (10) Pitch trim position;
- (11) Control column or pitch control surface position;
- (12) Control wheel or lateral control surface position;
- (13) Rudder pedal or yaw control surface position;
- (14) Thrust of each engine;
- (15) Position of each thrust reverser;
- (16) Trailing edge flap or cockpit flap control position; and
- (17) Leading edge flap or cockpit flap control position.

For the purpose of this section, *manufactured* means the point in time at which the airplane inspection acceptance records reflect that the airplane is complete and meets the FAA-approved type design data.

(e) After October 11, 1991, no person may operate a large airplane equipped with a digital data bus and ARINC 717 digital flight data acquisition unit (DFDAU) or equivalent unless it is equipped with one or more approved flight recorders that utilize a digital method of recording and storing data and a method of readily retrieving that data from the storage medium. Any parameters specified in appendix B of this part that are available on the digital data bus must be recorded within the ranges, accuracies, resolutions, and sampling intervals specified.

(f) After October 11, 1991, no person may operate an airplane specified in paragraph (b) of this section that is manufactured after October 11, 1991, nor an airplane specified in paragraph (a) of this section that has been type certificated after September 30, 1969, and

manufactured after October 11, 1991, unless it is equipped with one or more flight recorders that utilize a digital method of recording and storing data and a method of readily retrieving that data from the storage medium. The parameters specified in appendix B of this part must be recorded within the ranges, accuracies, resolutions, and sampling intervals specified.

(g) Whenever a flight recorder required by this section is installed, it must be operated continuously from the instant the airplane begins the takeoff roll until it has completed the landing roll at an airport.

(h) Except as provided in paragraph (i) of this section, and except for recorded data erased as authorized in this paragraph, each certificate holder shall keep the recorded data prescribed in paragraph (a), (b), (c), or (d) of this section, as appropriate, until the airplane has been operated for at least 25 hours of the operating time specified in §121.359(a). A total of 1 hour of recorded data may be erased for the purpose of testing the flight recorder or the flight recorder system. Any erasure made in accordance with this paragraph must be of the oldest recorded data accumulated at the time of testing. Except as provided in paragraph (i) of this section, no record need be kept more than 60 days.

(i) In the event of an accident or occurrence that requires immediate notification of the National Transportation Safety Board under part 830 of its regulations and that results in termination of the flight, the certificate holder shall remove the recording media from the airplane and keep the recorded data required by paragraph (a), (b), (c), or (d) of this section, as appropriate, for at least 60 days or for a longer period upon the request of the Board or the Administrator.

(j) Each flight recorder required by this section must be installed in accordance with the requirements of §25.1459 of this chapter in effect on August 31, 1977. The correlation required by §25.1459(c) of this chapter need be established only on one airplane of any group of airplanes --

(1) That are of the same type;

(2) On which the model flight recorder and its installation are the same; and

(3) On which there is no difference in the type design with respect to the installation of those first pilot's instruments associated with the flight recorder. The most recent instrument calibration, including the recording medium from which this calibration is derived, and the recorder correlation must be retained by the certificate holder.

(k) Each flight recorder required by this section that records the data specified in paragraph (a), (b), (c), or (d) of this section, as appropriate, must have an approved device to assist in locating that recorder under water.

(l) No person may operate an airplane specified in paragraph (b) of this section that meets

the Stage 2 noise levels of part 36 of this chapter and is subject to §91.801(c) of this chapter unless it is equipped with one or more approved flight data recorders that utilize a digital method of recording and storing data and a method of readily retrieving that data from the storage medium. The information specified in paragraphs (c)(1) through (c)(11) of this section must be able to be determined within the ranges, accuracies and recording intervals specified in appendix B of this part. In addition --

(1) This flight data recorder must be installed at the next heavy maintenance check after May 26, 1994, but no later than May 26, 1995. A heavy maintenance check is considered to be any time an aircraft is scheduled to be out of service for 4 or more days.

(2) By June 23, 1994, each carrier must submit to the FAA Flight Standards Service, Air Transportation Division (AFS-200), documentation listing those airplanes covered under this paragraph and evidence that it has ordered a sufficient number of flight data recorders to meet the May 26, 1995, compliance date for all aircraft on that list.

(3) After May 26, 1994, any aircraft that is modified to meet Stage 3 noise levels must have the flight data recorder described in paragraph (c) of this section installed before operating under this part.

[Doc. No. 24418, 52 FR 9636, Mar. 25, 1987, as amended by Amdt. 121-197, 53 FR 26147, July 11, 1988; Amdt. 121-238, 59 FR 26900, May 24, 1994]